



Ministry of Higher Education and  
Scientific Research - Iraq

Warith Al-Anbiyaa University  
College of Engineering  
Department of Aircraft Engineering



## MODULE DESCRIPTOR FORM

### نموذج وصف المادة الدراسية

Module Information			
معلومات المادة الدراسية			
<b>Module Title</b>	Aircraft Engines II محركات الطائرات II		<b>Module Delivery</b>
<b>Module Type</b>	CORE		Theory Lab Tutorial
<b>Module Code</b>	AIEN355		
<b>ECTS Credits</b>	6		
<b>SWL (hr/sem)</b>	150		
<b>Module Level</b>	3	<b>Semester of Delivery</b>	5
<b>Administering Department</b>	ME	<b>College</b>	ME
<b>Module Leader</b>	Dr. Aws Akram Mahmood AL_Akam	<b>e-mail</b>	<a href="mailto:aws@uowa.iq">aws@uowa.iq</a>
<b>Module Leader's Acad. Title</b>	Dr.Asst.Prof.Dr	<b>Module Leader's Qualification</b>	Ph.D.
<b>Module Tutor</b>		<b>e-mail</b>	
<b>Peer Reviewer Name</b>		<b>e-mail</b>	
<b>Review Committee Approval</b>	01/12/2025	<b>Version Number</b>	2025

Relation With Other Modules			
العلاقة مع المواد الدراسية الأخرى			
<b>Prerequisite module</b>	AIEN243	<b>Semester</b>	4
<b>Co-requisites module</b>	None	<b>Semester</b>	

## Module Aims, Learning Outcomes and Indicative Contents

أهداف المادة الدراسية ونتائج التعلم والمحتويات الإرشادية

<p><b>Module Aims</b> أهداف المادة الدراسية</p>	<ol style="list-style-type: none"> <li>1. Develop the knowledge of the basics concepts in constructing and modifying turbine aircraft engines.</li> <li>2. Awareness of theoretical concepts dealing with the operating cycle analysis and improving performance.</li> <li>3. Providing the knowledge in the engine testing under various conditions and identifying the evaluation parameters.</li> <li>4. Understanding the concepts of engagement between the various components of the turbine aircraft engines.</li> <li>5. Providing the knowledge of the different types of turbo-engines encountered in aviation sector.</li> </ol>
<p><b>Module Learning Outcomes</b> مخرجات التعلم للمادة الدراسية</p>	<ol style="list-style-type: none"> <li>1. Applying the concepts of thermodynamic cycles according to the second law of thermodynamics.</li> <li>2. Achieving the principle of energy conservation and the stage analysis for different turbine aircraft engines.</li> <li>3. Complete awareness of the determination techniques of each kind of turbine engines using basic formulation with testing procedures.</li> <li>4. Applying the first and second law concepts to the thermodynamic processes associated with the combustion systems in turbo-engines.</li> <li>5. Exploitation of the acquired knowledge in turbo-engines including description and cycle analysis of various kinds of engines involved.</li> </ol>
<p><b>Indicative Contents</b> المحتويات الإرشادية</p>	<p><b>Indicative contents include the following topics:</b></p> <p><u>Part A</u></p> <p><b>Introduction:</b> Aerostats, Aerodynes, Newton's laws of motion, Classification of turbo aircraft engines. <b>Turbine aircraft engines:</b> Turbine engine components, Intake/Diffuser, Compressor, Combustion chamber, Turbine, Nozzle. <b>Thermodynamic cycles:</b> Simple Brayton cycle, Brayton cycle with re-heat.[14hrs]</p> <p><b>Flow characteristics through turbomachine:</b> Compressibility effect, Flow momentum through a rotor. <b>Engine intake:</b> Flow in varying cross, section ducts, Subsonic intakes, Supersonic intakes, Gas dynamic relations for intakes</p> <p><b>Engine nozzle:</b> Types of nozzles, Flow characteristics in a nozzle, Gas dynamic relations for nozzles, Off-design operation of a nozzle. [14hrs]</p>

	<p><u>Part B</u></p> <p><b>Centrifugal machines:</b> Elementary theory of radial turbomachines, Centrifugal (Radial) compressor, Velocity diagrams, Stage analysis Preshirl, Slip, Rotating stall, Surging, Radial inflow turbine, Radial flow, analysis. [14hrs]</p> <p><b>Axial flow machines:</b> Elementary theory of axial flow turbomachines, Blade terminology, Axial flow compressor, Velocity diagrams, Stage analysis of axial compressor, Characteristics of axial compressor , Axial, flow turbine, Velocity diagrams, Stage analysis of axial turbine, Characteristics of axial turbine. [15hrs]</p> <p><b>Combustion system:</b> Types of aviation fuel, Theory of combustion, Calorific value of fuel, Combustion in turbine engines, Exhaust gas analysis. <b>Aircraft engines:</b> Propulsion efficiency , Thrust producing engines, Turbojet engine, Turbofan engine, Afterburning , Power producing engines, Turboprop engine, Turboshaft engine [15hrs]</p>
<p style="text-align: center;"><b>Learning and Teaching Strategies</b></p> <p style="text-align: center;">استراتيجيات التعلم والتعليم</p>	
<p><b>Strategies</b></p>	<ul style="list-style-type: none"> <li>• Deliver well-structured lectures that explain classification of aero-engines performance, the thermodynamic cycles, flow through turbomachines and the main parts of simple turbojet engine including intakes, compressor (axial and centrifugal) , combustion chamber, turbine, and nozzle. Utilize clear visuals like diagrams, animations, and real-world examples to enhance understanding.</li> <li>• Incorporate active learning activities within lectures. This could involve short quizzes, interactive questions, group discussions, or problem-solving exercises to solidify understanding and encourage student participation.</li> <li>• Provide students with a variety of problem-solving exercises, ranging from introductory to more challenging. This caters to different learning styles and allows students to build confidence as they progress.</li> <li>• Whenever possible, connect the theoretical concepts to real-world engineering applications of gas dynamics.</li> <li>• Incorporate regular quizzes and homework assignments to assess student understanding and identify areas needing improvement. Provide timely and constructive feedback to guide student learning.</li> <li>• Utilize well-designed midterm and final exams that test both theoretical knowledge and problem-solving abilities in aircraft engines.</li> </ul>

## Student Workload (SWL)

الحمل الدراسي للطالب

<b>Structured SWL (h/sem)</b> الحمل الدراسي المنتظم للطالب خلال الفصل	78	<b>Structured SWL (h/w)</b> الحمل الدراسي المنتظم للطالب أسبوعياً	5
<b>Unstructured SWL (h/sem)</b> الحمل الدراسي غير المنتظم للطالب خلال الفصل	72	<b>Unstructured SWL (h/w)</b> الحمل الدراسي غير المنتظم للطالب أسبوعياً	4.8
<b>Total SWL (h/sem)</b> الحمل الدراسي الكلي للطالب خلال الفصل	150		

## Module Evaluation

تقييم المادة الدراسية

		Time/ Number	Weight (Marks)	Week Due	Relevant Learning Outcome
<b>Formative assessment</b>	<b>Quizzes</b>	4	20% (20)	3, 6,9,12	All
	<b>Assignments</b>	2	10% (10)	5, 10	All
	<b>Projects / Lab.</b>	Lab. 4	10% (10)	Continuous	
	<b>Report</b>	-	-	-	-
<b>Summative assessment</b>	<b>Midterm Exam</b>	2 hrs.	10% (10)	7	All
	<b>Final Exam</b>	3 hrs.	50% (50)	16	All
<b>Total assessment</b>			100% (100 Marks)		

## Delivery Plan (Weekly Syllabus)

المنهاج الأسبوعي النظري

	Material Covered
<b>Week 1</b>	Introduction to the different types of aircraft engines
<b>Week 2</b>	Description of the main components of the turbine aircraft engines
<b>Week 3</b>	Description of Brayton standard cycle and Brayton cycle with afterburning.
<b>Week 4</b>	Describing the flow characteristics through turbomachines.
<b>Week 5</b>	Description, classification and analysis of aircraft engine intakes.
<b>Week 6</b>	Description, classification and analysis of aircraft engine nozzles.
<b>Week 7</b>	Elementary theory of radial turbomachinery.
<b>Week 8</b>	Stage analysis and velocity diagrams involved in centrifugal compressor.
<b>Week 9</b>	Performance characteristics of centrifugal compressors and turbines.
<b>Week 10</b>	Elementary theory of axial flow turbomachinery.
<b>Week 11</b>	Stage analysis and performance characteristics for axial flow compressors.
<b>Week 12</b>	Stage analysis and performance characteristics for axial flow turbines.

<b>Week 13</b>	Combustion system description and analysis in turbo-aircraft engines.
<b>Week 14</b>	Description and analysis of the thrust producing aircraft engines (turbojet and turbofan engines).
<b>Week 15</b>	Description and analysis of the power producing aircraft engines (turboprop and turboshaft engines).
<b>Week 16</b>	<b>Final Exam</b>

<b>Delivery Plan (Weekly Lab. Syllabus)</b> المنهاج الأسبوعي للمختبر	
	<b>Material Covered</b>
<b>Week 1</b>	Exp. 1: Turbo jet engine
<b>Week 2</b>	Exp. 2: Turbo fan engine
<b>Week 3</b>	Exp. 3: Turboprop and Turboshaft engines
<b>Week 4</b>	Exp. 4: Two shaft gas turbine unit
<b>Week 5</b>	Exp. 5:
<b>Week 6</b>	Exp. 6:
<b>Week 7</b>	Exp. 7:

<b>Learning and Teaching Resources</b> مصادر التعلم والتدريس		
	<b>Text</b>	<b>Available in the Library?</b>
<b>Required Texts</b>	H.I.H. Saravanamnttoo, et al. "Gas Turbine Theory" 7 <sup>th</sup> edition, Pearson Education Limited, 2017.	Yes
<b>Recommended Texts</b>	S.L. Dixon and C.A. Hall, "Fluid Mechanics and Thermodynamics of Turbomachinery".7 <sup>th</sup> edition, Butterworth-Heinemann, Elsevier, 2014.	No
<b>Websites</b>		

## APPENDIX:

GRADING SCHEME مخطط الدرجات				
Group	Grade	التقدير	Marks (%)	Definition,
Success Group (50 - 100)	A - Excellent	امتياز	90 - 100	Outstanding Performance
	B - Very Good	جيد جدا	80 - 89	Above average with some errors
	C - Good	جيد	70 - 79	Sound work with notable errors
	D - Satisfactory	متوسط	60 - 69	Fair but with major shortcomings
	E - Sufficient	مقبول	50 - 59	Work meets minimum criteria
Fail Group (0 – 49)	FX – Fail	مقبول بقرار	(45-49)	More work required but credit awarded
	F – Fail	راسب	(0-44)	Considerable amount of work required
Note:				
NB Decimal places above or below 0.5 will be rounded to the higher or lower full mark (for example a mark of 54.5 will be rounded to 55, whereas a mark of 54.4 will be rounded to 54. The University has a policy NOT to condone "near-pass fails" so the only adjustment to marks awarded by the original marker(s) will be the automatic rounding outlined above.				